

FOSSEE Fellowship Report

on

FILM COOLING ON A FLAT PLATE BY AIR-WATER MIST INJECTION

Submitted by JISHNU HANDIQUE

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Nomenclature

S	Slot height, m	
$d_{\rm w}$	Water droplet diameter, m	
q ^{//}	Heat flux, W/m ²	
U	Velocity, m/s	
T_h	Temperature of hot gas, K	
Tc	Temperature of coolant, K	
T_{f}	Temperature after coolant injection, K	
Ν	Phase	
Ι	Mass transfer rate, kg/s	
i, j, k	Indices	
i, j, k g	Indices Gravitational acceleration, m/s ²	
i, j, k g F	Indices Gravitational acceleration, m/s ² Interactive force per unit volume, N/m ³	
i, j, k g F p	Indices Gravitational acceleration, m/s ² Interactive force per unit volume, N/m ³ Pressure, Pa	
i, j, k g F p Q	Indices Gravitational acceleration, m/s ² Interactive force per unit volume, N/m ³ Pressure, Pa Rate of heat transfer per unit mass, W/kg	
i, j, k g F p Q W	Indices Gravitational acceleration, m/s ² Interactive force per unit volume, N/m ³ Pressure, Pa Rate of heat transfer per unit mass, W/kg Rate of work done per unit mass, J/kg	
i, j, k g F Q W e*	Indices Gravitational acceleration, m/s ² Interactive force per unit volume, N/m ³ Pressure, Pa Rate of heat transfer per unit mass, W/kg Rate of work done per unit mass, J/kg Total internal energy per unit mass, J/kg	
i, j, k g F Q W e* C _p	Indices Gravitational acceleration, m/s ² Interactive force per unit volume, N/m ³ Pressure, Pa Rate of heat transfer per unit mass, W/kg Rate of work done per unit mass, J/kg Total internal energy per unit mass, J/kg	

Greek Symbols

θ	Secondary coolant injection angle, ⁰
α	Volume fraction
μ	Dynamic viscosity, Pa.sec
σ_{c}	Phase stress tensor, Pa
ξ	Energy interaction term

Notations

$\delta/\delta t$	Partial time derivative
$\delta/\delta x$	Partial positional derivative

Chapter1 Introduction and Problem Statement

1.1. Introduction

Film cooling is mainly used in gas-turbine operation. A low-temperature secondary fluid is injected to the surface exposed to high temperature gas. The coolant fluid forms a film over the surface and protects it from the hot gas [1]. This process is known as film cooling.



Figure 1.1(a). Film cooling in a gas turbine blade [2]



Figure 1.1(b). Cad model of film cooling holes in turbine blade [3]

1.2. Problem Statement

A film cooling problem on a flat surface was simulated by using a 2D model. The secondary hole was assumed as slotted hole with height 4 mm. The temperature of main stream fluid air, T_g and the secondary fluid air-water mist, T_c were 400K and 300K respectively. Secondary coolant fluid was injected at 35⁰. Main stream velocity, U_g = 10 m/sec and U_c = 10 m/sec. The simulations were carried out for mist loading fraction, f = 2%, 5%, 15% and 25%. thermalPhaseChange phase model was implemented to capture the phase change due to temperature. A multiphase solver reactingMultiphaseEulerFoam was used to study the problem [4]. The detailing of the geometry was shown clearly in the Figure1.2.



Figure 1.2. 2D Grid

Parameter	Detail
Model	2 Dimensional
Geometry-Mesh creating software	ICEM CFD
Number of cells	11,699
Post-processing tool	Paraview, Sigma Plot
Solver	reactingMultiphaseEulerFoa m
Turbulence model	Standard k– ϵ
Pressure-velocity coupling	PIMPLE algorithm [4]
Convective term solving scheme	Gauss upwind [4]
Turbulent term solving scheme	Gauss upwind [4]

Table 1. Geometry and Computational Details

Parameter	Value/Condition
µ water	3.645e-05 Pa.sec
(C _p) _{water}	4195 J/kgK
Pr _{water}	2.289
s	0.004 m
θ	35 ⁰
q ^{//}	0 W/m^2
T _h	400 K
T _c	300 K
U _h (Hot)	10 m/sec
U _{air} (Coolant)	10 m/sec
Uwater (Coolant)	0.6 m/sec

Table 2. Fluid properties and initial conditions

Chapter2 Equations

2.1. Individual Phase Continuity Equation [5]

$$\frac{\partial}{\partial t}(\rho_N\alpha_N) + \frac{\partial}{\partial x_i}(\rho_N\alpha_N U_{N_i}) = I_N$$

2.2. Individual Phase Momentum Equation [5]

$$\frac{\partial}{\partial t} \left(\rho_N \alpha_N U_{N_k} \right) + \frac{\partial}{\partial x_i} \left(\rho_N \alpha_N U_{N_i} U_{N_k} \right) = \alpha_N \rho_N g_k + F_{N_k} - \delta_N \{ \frac{\partial p}{\partial x_k} - \frac{\partial \sigma^D c_{ki}}{\partial x_i} \}$$

2.3. Individual Phase Energy Equation [5]

$$\frac{\partial}{\partial t}(\rho_{N}\alpha_{N}e^{*}{}_{N}) + \frac{\partial}{\partial x_{i}}(\rho_{N}\alpha_{N}e^{*}{}_{N}U_{N_{i}}) = Q_{N} + W_{N} + \xi_{N} - \delta_{N}\frac{\partial}{\partial x_{j}}(U_{C_{i}}\sigma_{C_{ij}})$$

2.4. Effectiveness

$$\eta = \frac{T_h - T_f}{T_h - T_c}$$

Chapter3 Results and Discussion

3.1. Validation

The effectiveness was found out over flat plate from the secondary inlet position. Then the outcomes were validated with the numerical work of Li and Wang [6].



Figure 1.3. Validation

3.2. Comparison Between Results of Air and Air-Water Mist as Coolant



Figure 1.4. Comparison of Effectiveness for No Mist (f = 0) and Mist Injection (f = 5%)

3.3. Contours



Figure 1.5. Temperature contour (f = 0)



Figure 1.6. Temperature contour (f = 5%)

3.4. Conclusion

From the numerical works we can conclude that the injection of mist protects the flat surface from hot gases better than the air injection system.

Reference

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